

Space Hazards Applications, LLC

Building Commercial Applications To Safeguard Satellites From Space Weather Hazards

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Space Hazards Applications, LLC



Our goal is to create a suite of easy to access applications that give the unique real time and retrospective space weather hazard to specific satellite systems along any satellite orbit. Assess and mitigate on orbit space weather hazards to satellites from the Earth to moon

- Real-time Models
- Web Applications
- Consulting Services



Particle Radiation Hazards

Regions

SPIS (Roussel et al., 2005)	Surface Charging: Charged particles collect on satellite surfaces producing high differential voltages, electrostatic discharges, and electromagnetic interference.	Low energy protons /electrons	Credit:NASA
Schneider et al., 2016	Internal Charging: Energetic electrons accumulate in dielectrics (circuit boards, cable insulators) and on ungrounded metal (spot shields, connector contacts) leading to damaging discharges.	High energy electrons	Magnetosphere
N+ N+ https://semiwiki.com/ee	Single Event Upsets: Energetic ion passage through microelectronic device node causes catastrophic device failure, latent damage, or uncommanded mode/state changes.	High energy protons/ions	
1 1 1 1 1 1 1 1 1 1 1 1 1 1	Total Dose: Energy loss when proton/electrons pass through microelectronic devices causing degradation and reduced performance that accumulates over mission (or step-wise during high dose rate events).	High energy protons electrons	Moon

LEO Space Traffic Coordination

- ~6700 satellites in orbit, 5938 in Low Earth Orbit
- Number of Conjunction Data Messages (CDMs) sent to warn operators of possible collisions is increasing
- 27 "emergency" conjunction messages were sent daily in 2019
- Space weather is an overlooked aspect of Space Traffic Coordination
- An issue for one satellite can escalate to an issue for many (i.e Galaxy 15 zombiesat)



Source: 18th Space Control Squadron (DoD)



Single Event Effects: SPAM Model

Solar Particle Access Model (SPAM)



An empirically derived data driven model for mapping solar particle flux throughout the magnetosphere for satellite anomaly monitoring and attribution

Solar Energetic Particles

Solar energetic particle (SEP) events occur when the sun launches a fast moving coronal mass ejection (CME).

Ions are accelerated in the shock front preceding the CME. They stream ahead and flood near Earth space for days to weeks.

SEP stats

- 270 events impacted Earth since 1976 (based on GOES 10 MeV protons >10 /cm2-s-str)
- 0-23 per year, average ~ 6
- Occurrence rate follows the solar cycle but large events can happen any time
- 3 in 2023





SOLAR PARTICLE ACCESS

Some regions are shielded as ions are deflected by Earth's magnetic field.

Polar orbiting MEO/LEO satellites will pass in and out of high flux regions.

Monitoring the threat from these ions requires knowledge of where they have access or are deflected.

Access regions change as the magnetic field is distorted by the oncoming solar wind.



Single Event Effects: SPAM Model

POES proton flux



SPAM defines SEP access using real time observations of ions in Low Earth Orbit (POES/MetOP)

Empirically maps each POES/MetOp pass through the cutoff regions to:



All Magnetic Local times

All altitudes

All ion species

Translates to LET spectrum and SEU rate

SatCAT/SPAM

Easy to use online interface

Satellite Charging Assessment Tool (SatCAT)/ Solar Particle Access Model (SPAM)

	COLLECTIONS	ADD AN	OMALY LISTS	ADD P/	ARTS FIXED GEO	
New Collection Parameters Create a new dataset by setting the parameters and clicking 'Create Dataset'. Choose a satellite trajectory from the drop down list or add a new one with the [+]. An		Available Data Collections Below are descriptions of datasets available for analysis and display. Four at fixed longitudes are openly available to all users. All other are accessible data. To delete your datasets click the [x] in the last column.				
mail will be sent when gene complete and data is ready for nay take several hours to ge	ration is or viewing. (It nerate a year of	Name	Satellite	Туре	Parameters	
Collection Name		GEO 90 degree E long		Internal Charging	IC: Kapton (298- 473 K), mils	
Start Time: 2022-07-30 02	:15:06	GEO 180 degree E long		Internal Charging	IC: Kapton (298- 473 K), mils	
Real-time: 2022-07-30 02:	15:06	GEO 360 degree E long		Internal Charging	IC: Kapton (298- 473 K), mils	
nternal Charging: 🔲 Naterials: 🔔		Van Allen B	Van Allen B	Internal Charging	IC: Teflon generic (273 K), 5 20 100 200 mile	

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Solar Particle Options									
O None	Species	⊖ let	O SEU						
Thickness:	e.g. 50 (mils) 20	_							
Species Flux									

anergies: O h+ O all species butput inergies: 1.0,5.0,10.0 (MeV/nuc)

LET Flux

LET (MeV/(g/cm2)): 1.0,5.0,10.0 etc Material: None
SEU Rate SEU Parts: Add new part: + CREATE DATASET

- Create a dataset by filling in selections
- Choose a satellite
- Add start and stop time (or real time updates)
- Add shielding layers
- Choose to view particle flux, LET flux, or SEU rate (select from a list of common parts or add your own)
- The tool collects TLE's, creates the trajectory, gets the ion flux from SPAM, translates to LET and SEU for specific component and and stores in a database for display and download

SATCAT/SPAM



Visualization

GUI allows you to plot data (such as SEU rate) along with anomalies to identify correlations

Analysis

Determine whether your system is susceptible to ion impacts and monitor for future events

Example using SPAM to show the SEU rate along the NOAA 18 satellite orbit (polar at ~850 km) for a generic device

SUMMARY

We develop commercial models and applications for understanding unique real time space weather impacts to satellite systems.

<u>SEE</u>

- SPAM model, SPAM/SatCAT application
 Internal charging
- SHELLS-hires model, SatCAT application
 <u>Surface charging (coming soon)</u>
- Scopuli environment model
- Lunar Orbit (coming soon)
- surface charging environment and impacts

