

The effect of aerodynamic modeling on accelerometer transplant accuracy

Frederik Jacobs, Christian Siemes



Accelerometer issues of GRACE/-FO

- Accelerometers measure the non-gravitational accelerations, so that the effect can be removed from the ranging measurements
 - Accelerometer issues
 - **GRACE-B:** Turned off in 2016 to save power
 - **GRACE-D:** Noise levels increased a few weeks after launch
 - With one accelerometer malfunctioning or being turned off, time-variable gravity field models are (much) less accurate
- Solution? Transplant!
- Modeled accelerometer product to replace the malfunctioning accelerometer
 - Developed since 2006 (Save, 2006)

Transplants

Existing transplants

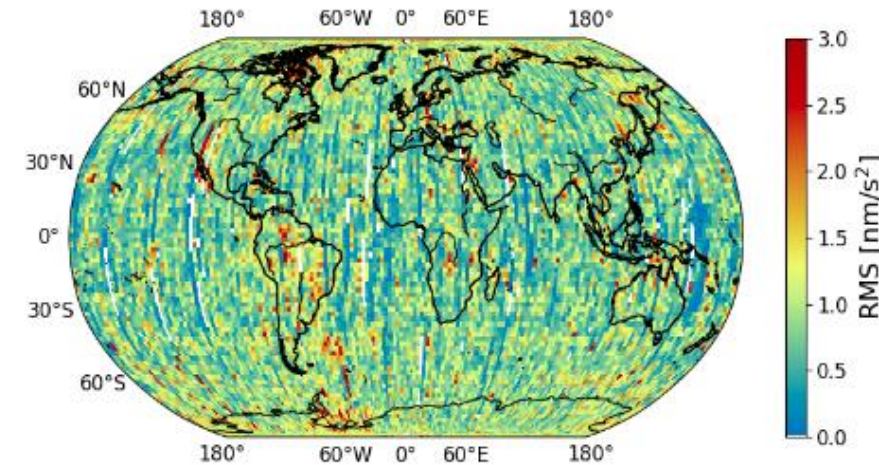
- JPL (Bandikova ,2019; Harvey, 2022; Harvey, 2024)
- ZARM (University of Bremen) (Huckfeldt, 2023)
- TU Graz (Behzadpour, 2019)
- Leibnitz University Hannover (Romeshkani, 2025)

Each transplant has its own methodology and assumptions

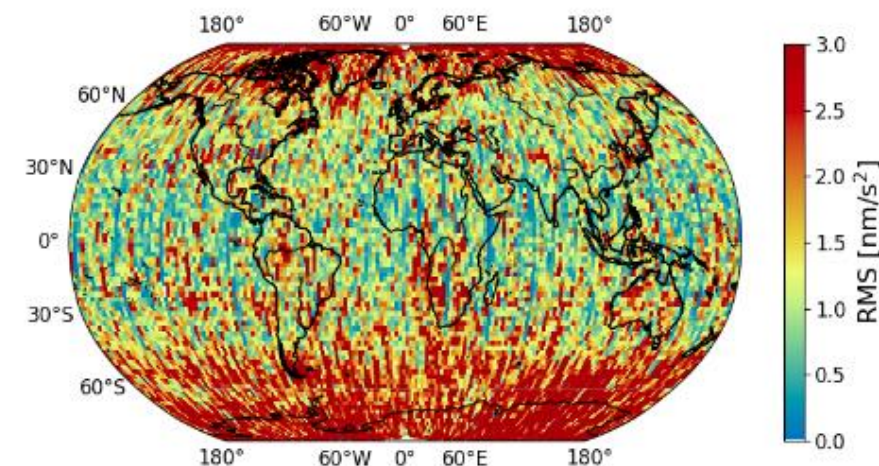
- Attitude correction
- Aerodynamic/radiation/thruster models
- Machine learning

→ But increased post-fit residual RMS

January 2003: before transplant era

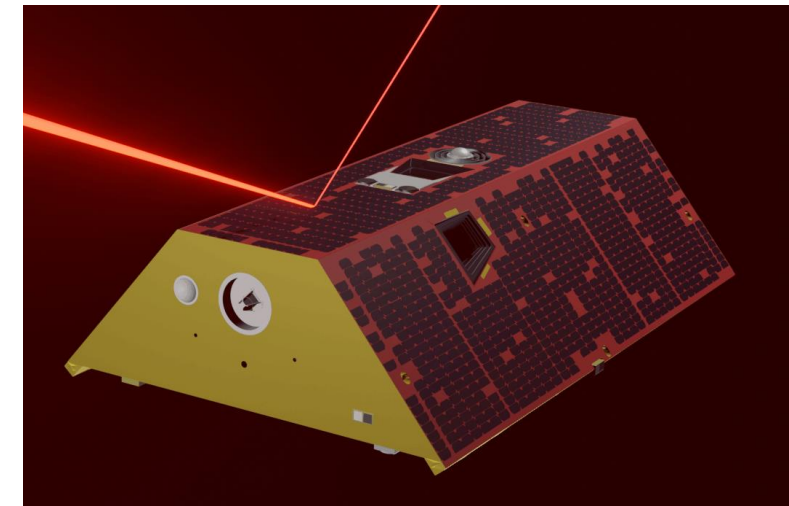
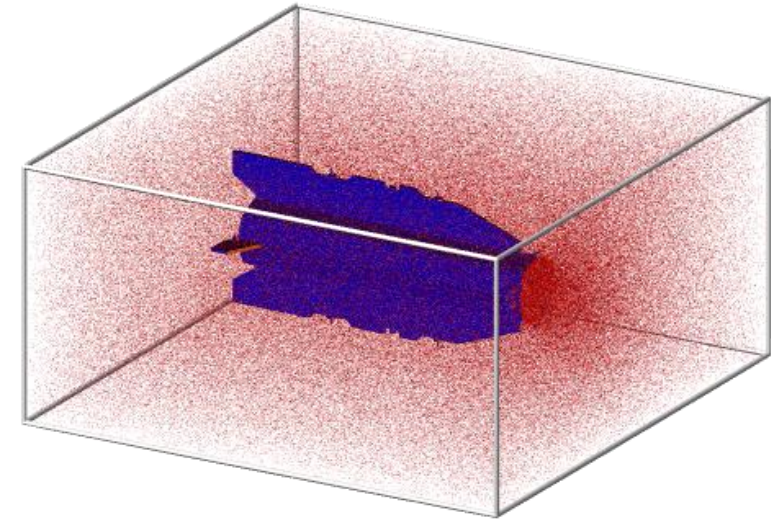


January 2017: transplant era



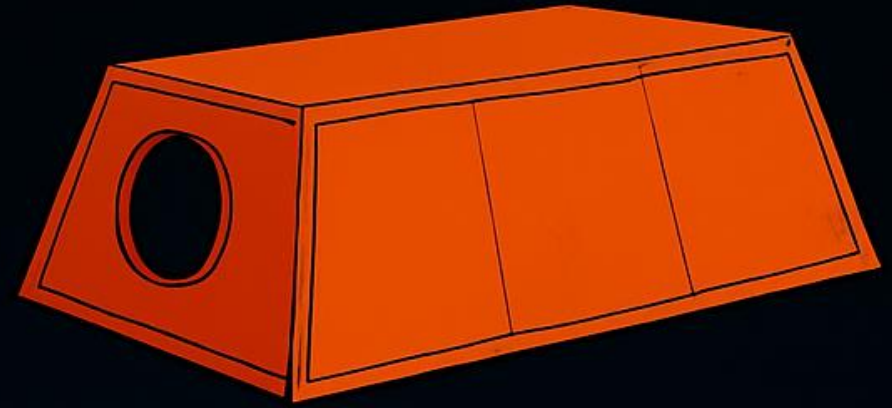
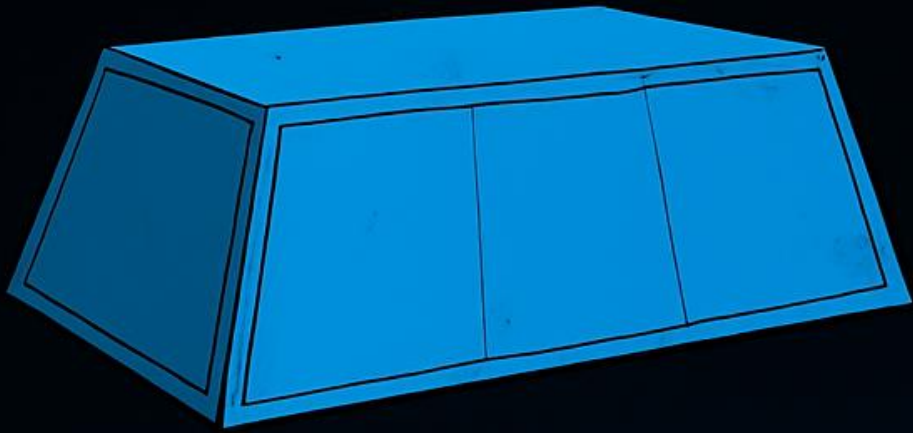
Why a TU Delft transplant?

- Expertise in density processing: high-fidelity modeling
- Aerodynamics:
 - DSMC aerodynamic force coefficients (March, 2019)
 - Cross-wind and density retrieval (Doornbos, 2010)
- Radiation pressure (Hładczuk, 2024)
 - Ray-tracing radiation tables
 - Thermal inertia
- Could lift the accuracy of the transplant to a higher level



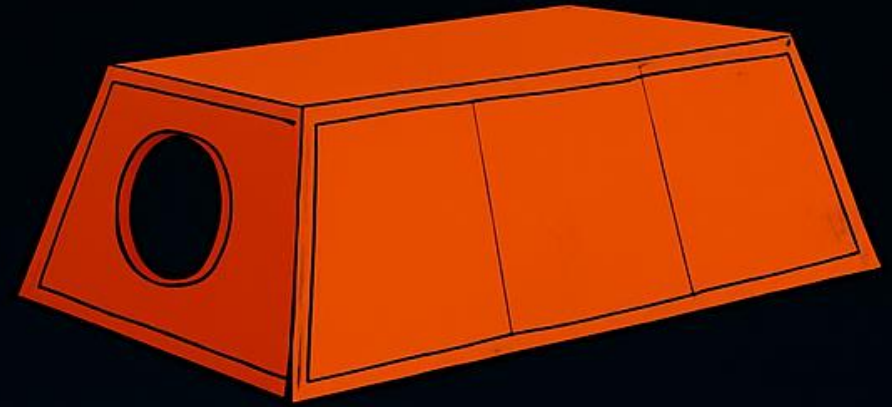
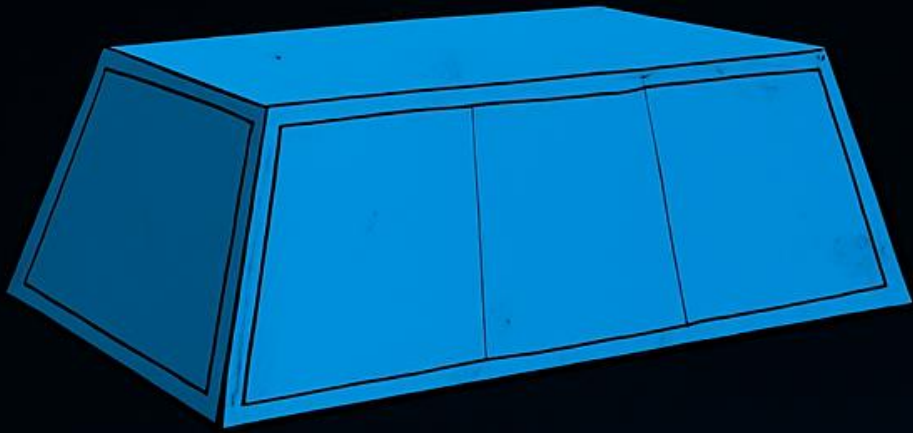
Transplant methodology

Acceleration → Density and wind



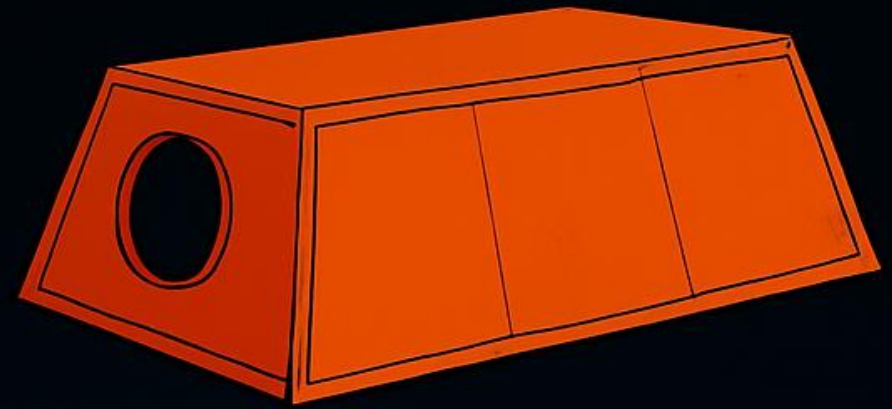
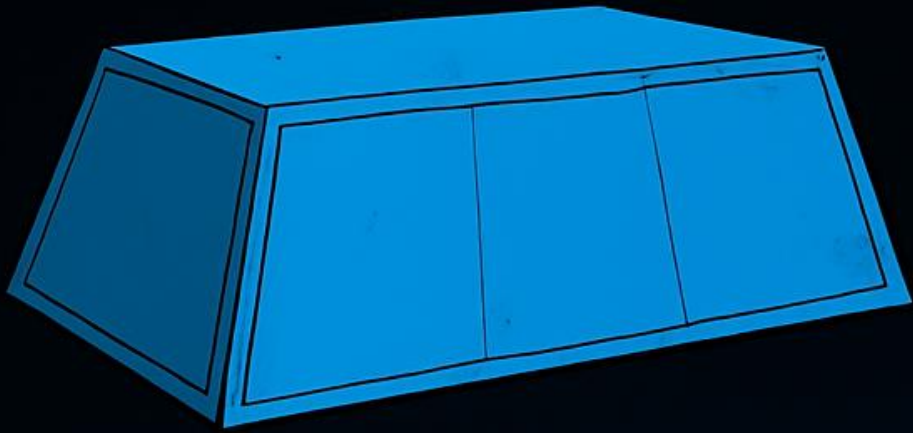
Transplant methodology

Acceleration → Density and wind → Density and wind → Acceleration



Transplant methodology

Density and wind → Acceleration



How can we validate transplants?

Need to know the accuracy of the transplant

Typically, transplants are only **produced when one accelerometer is down**

→ It cannot be compared to the other accelerometer

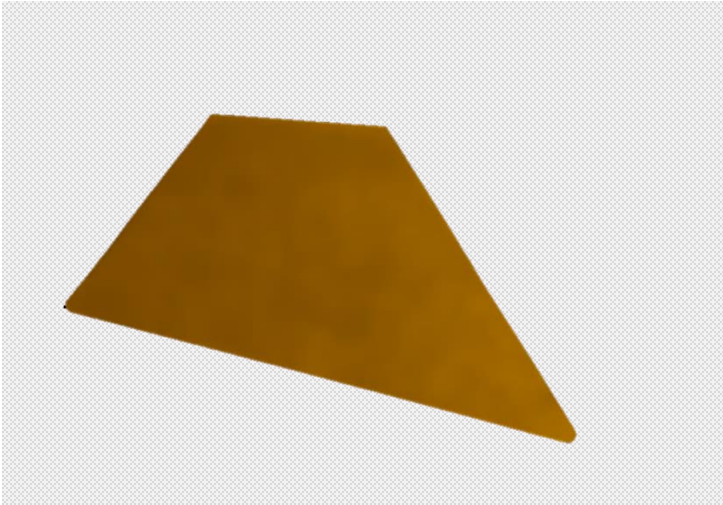
Select two reference years for GRACE for which both accelerometers were operational

- 2003: high solar activity; aerodynamically dominated
- 2009: low solar activity; radiation pressure dominated

Compare the transplant (from GRACE-A to GRACE-B) to the accelerometer (GRACE-B)

This presentation focuses on aerodynamic modeling → 2003

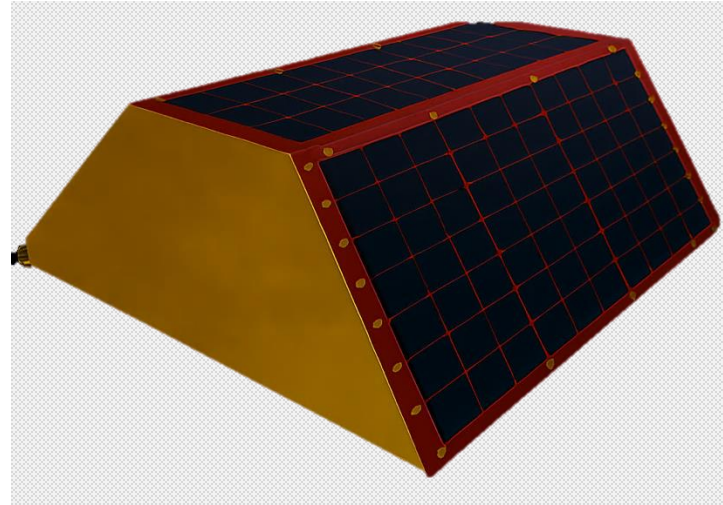
Aerodynamic models



Constant drag coefficient (2.4)

Drag

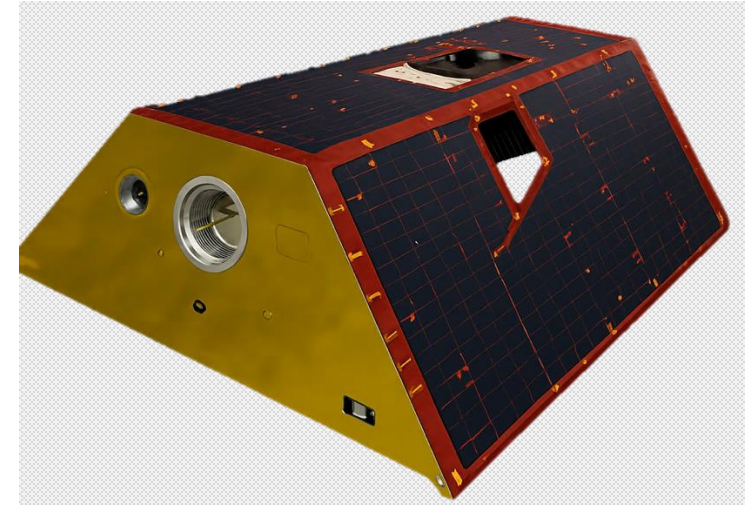
$$\vec{F}_D = \frac{1}{2} \rho v C_D A_{proj} \vec{v}$$



Panel model (Sentman, 1961)

Drag and lift

$$\vec{F}_D = \sum_i \frac{1}{2} \rho v^2 A_i (C_{D,i} \vec{e}_D + C_{L,i} \vec{e}_{L,i})$$

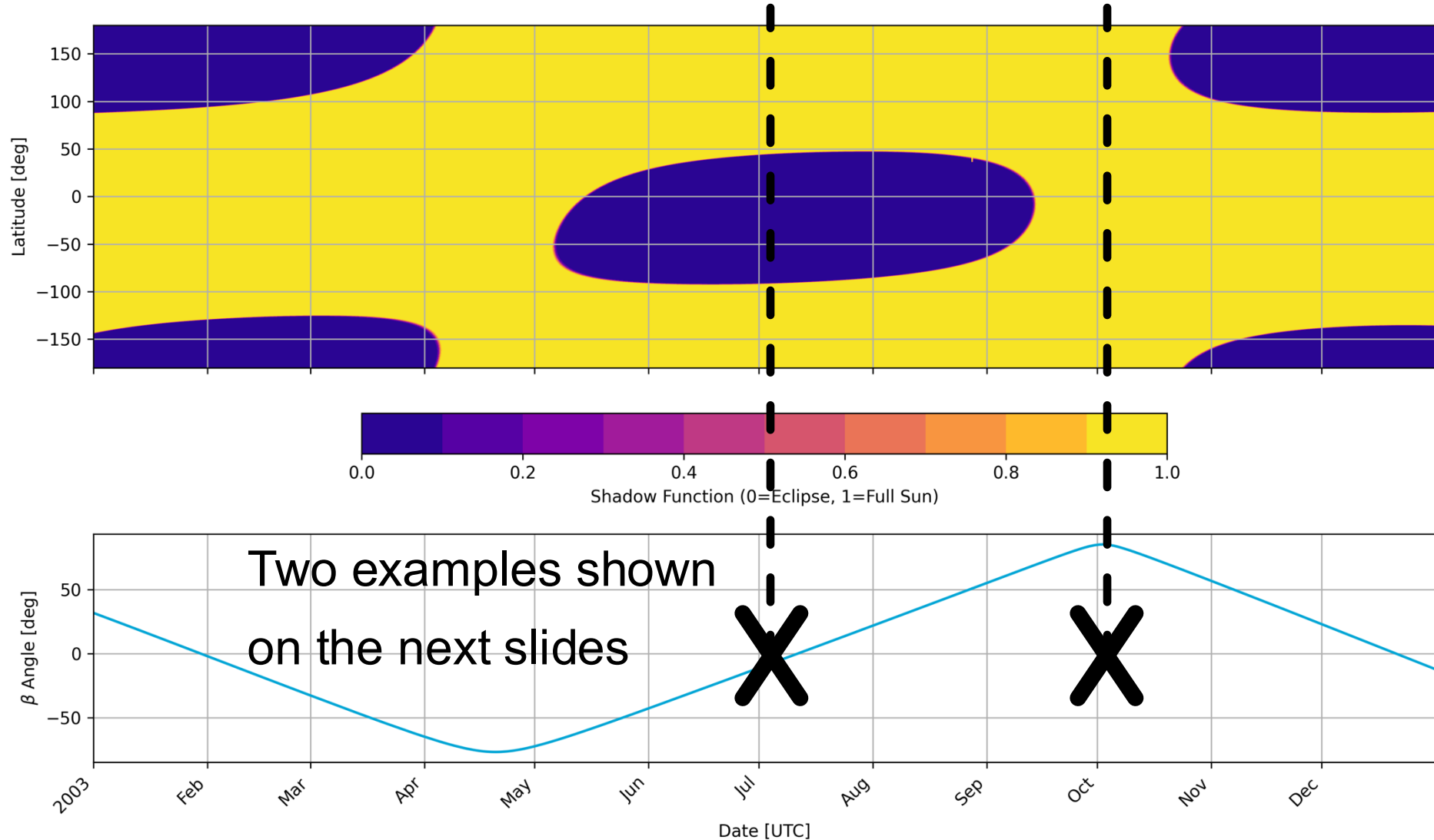


DSMC simulations (Bird, 1994)

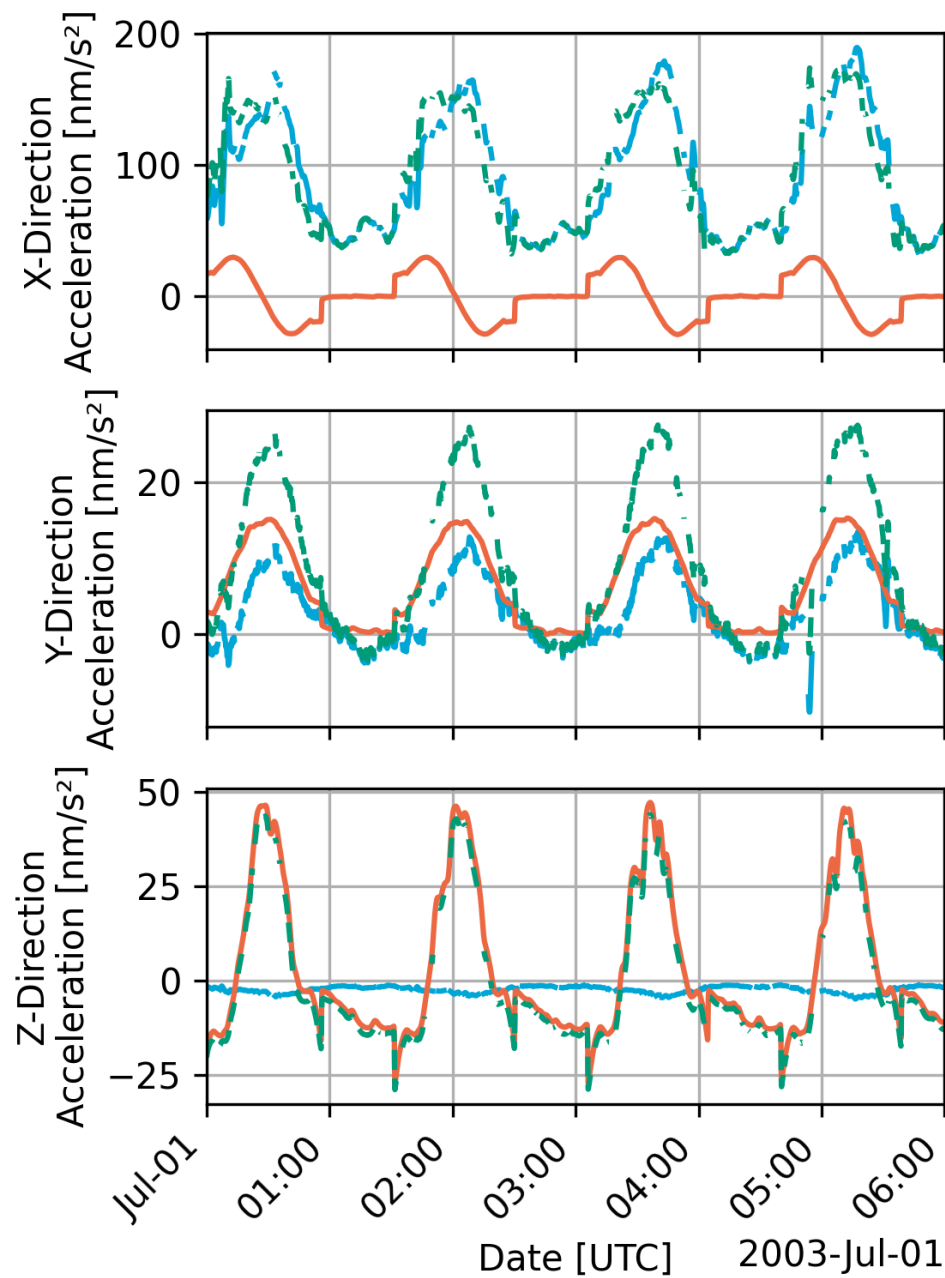
Drag, lift, multiple reflections and shadowing

$$\vec{F}_D = \frac{1}{2} \rho v^2 A_{ref} \vec{C}$$

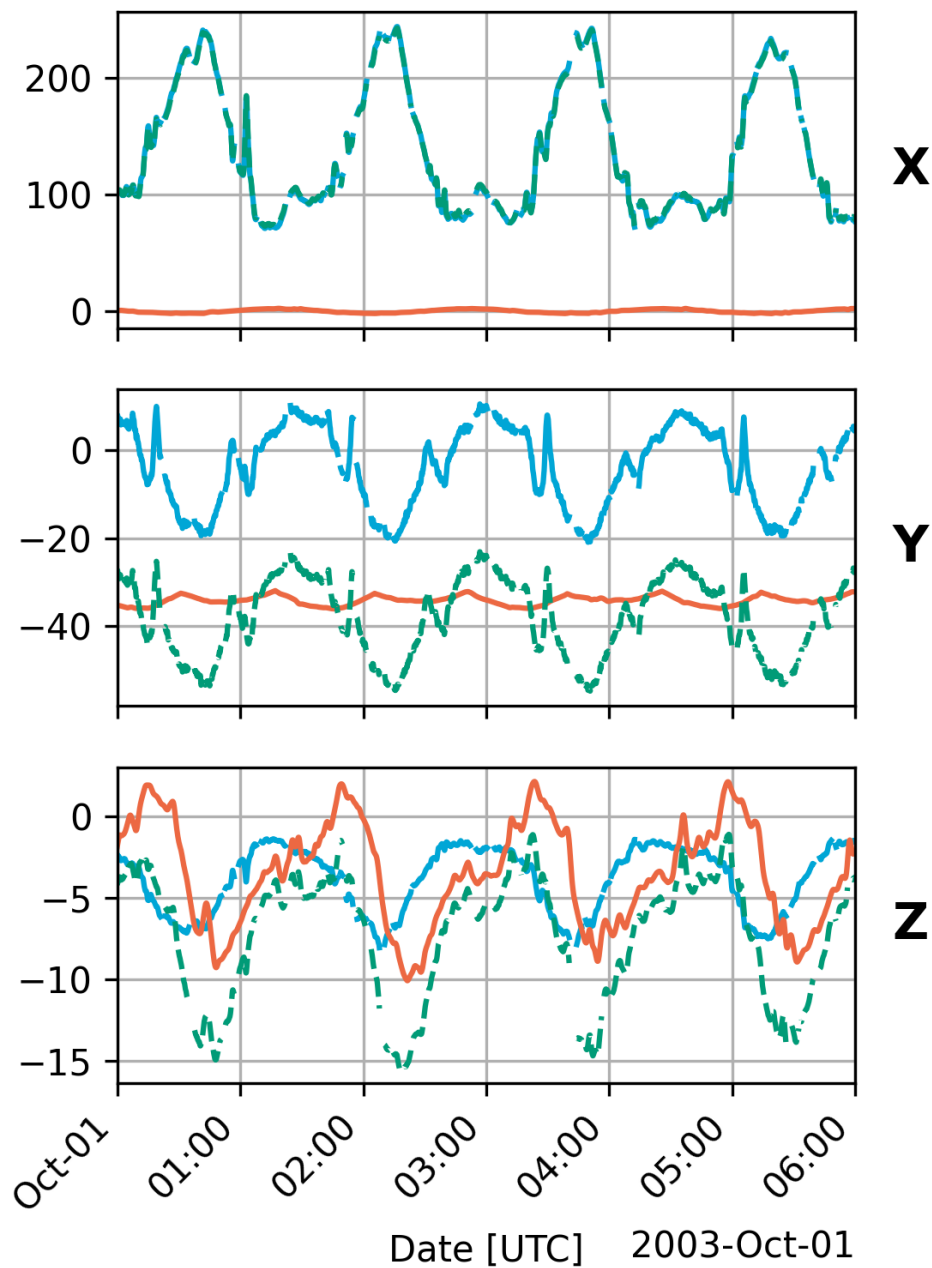
Shadow function and β angle sun



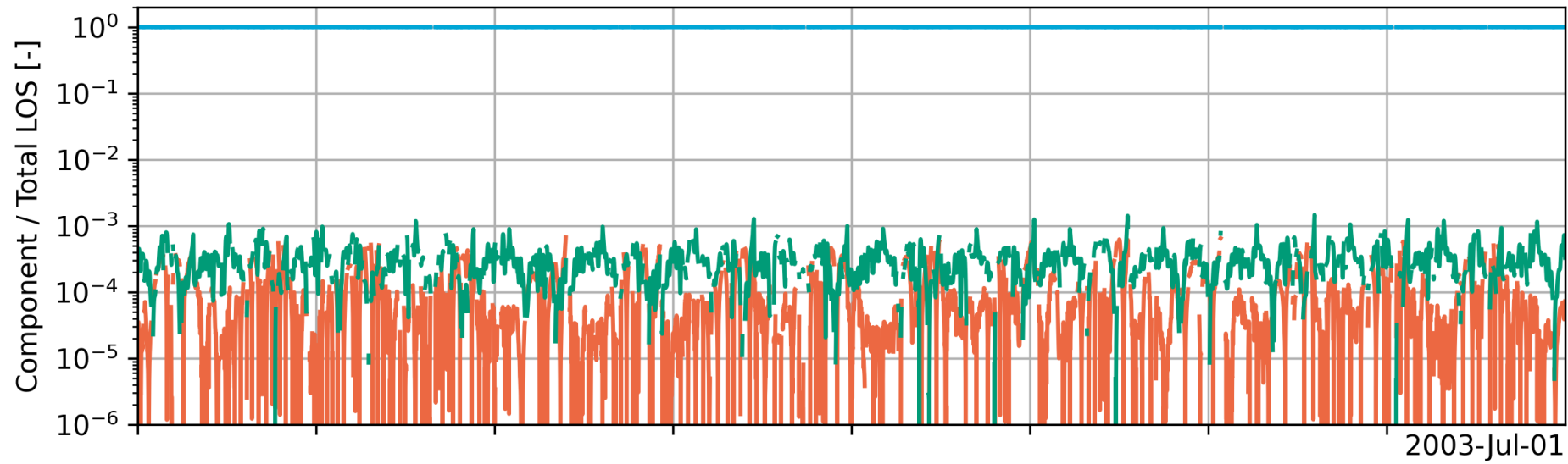
2003-07-01



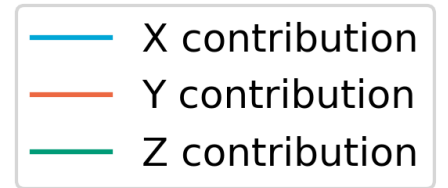
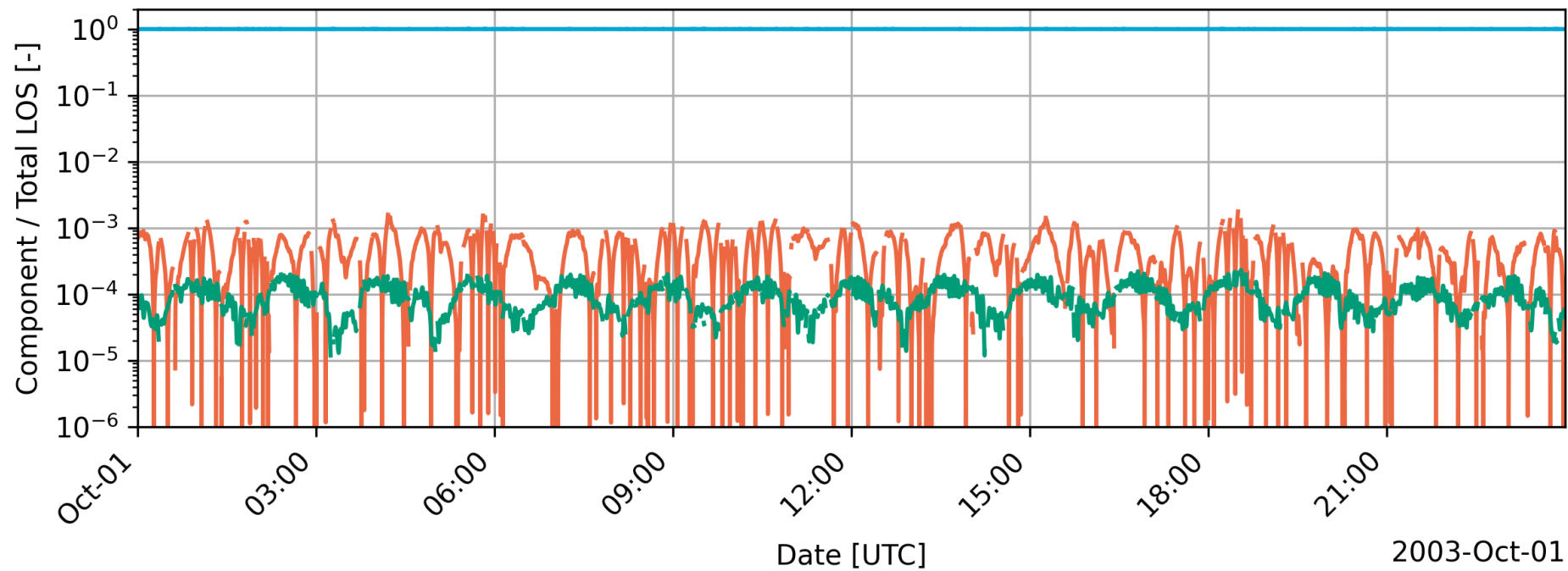
2003-10-01



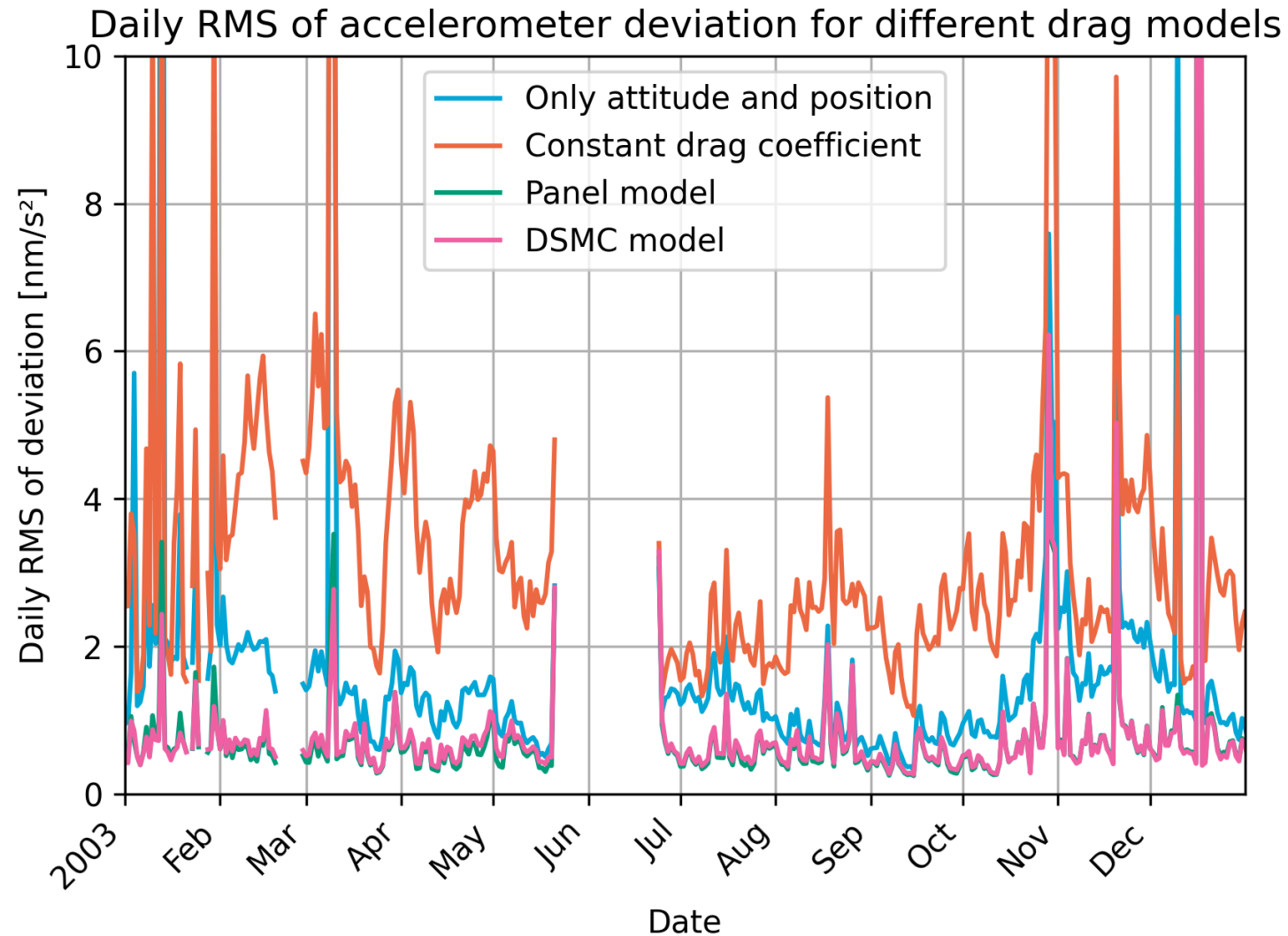
Directional Contribution on 2003-07-01



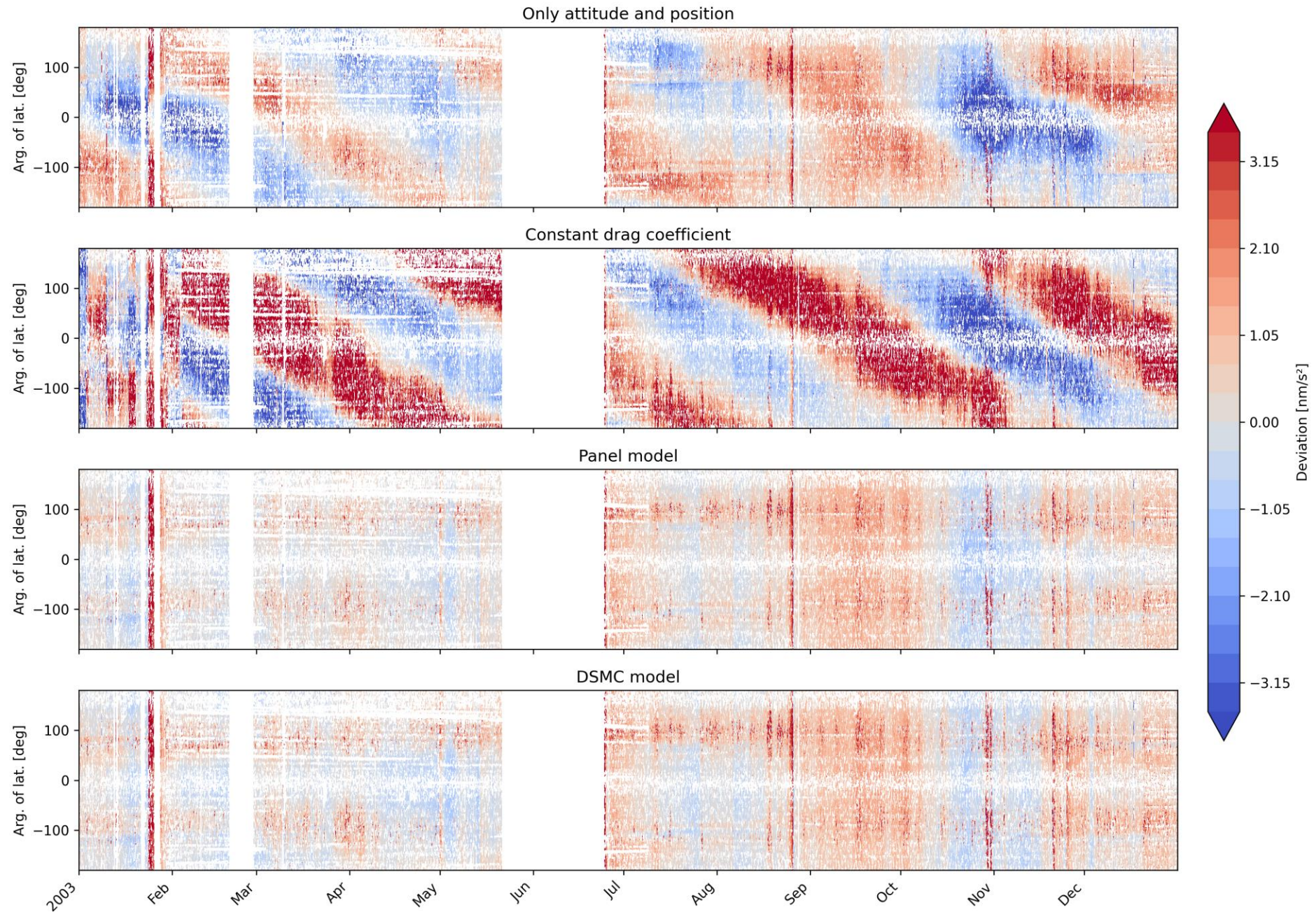
Directional Contribution on 2003-10-01



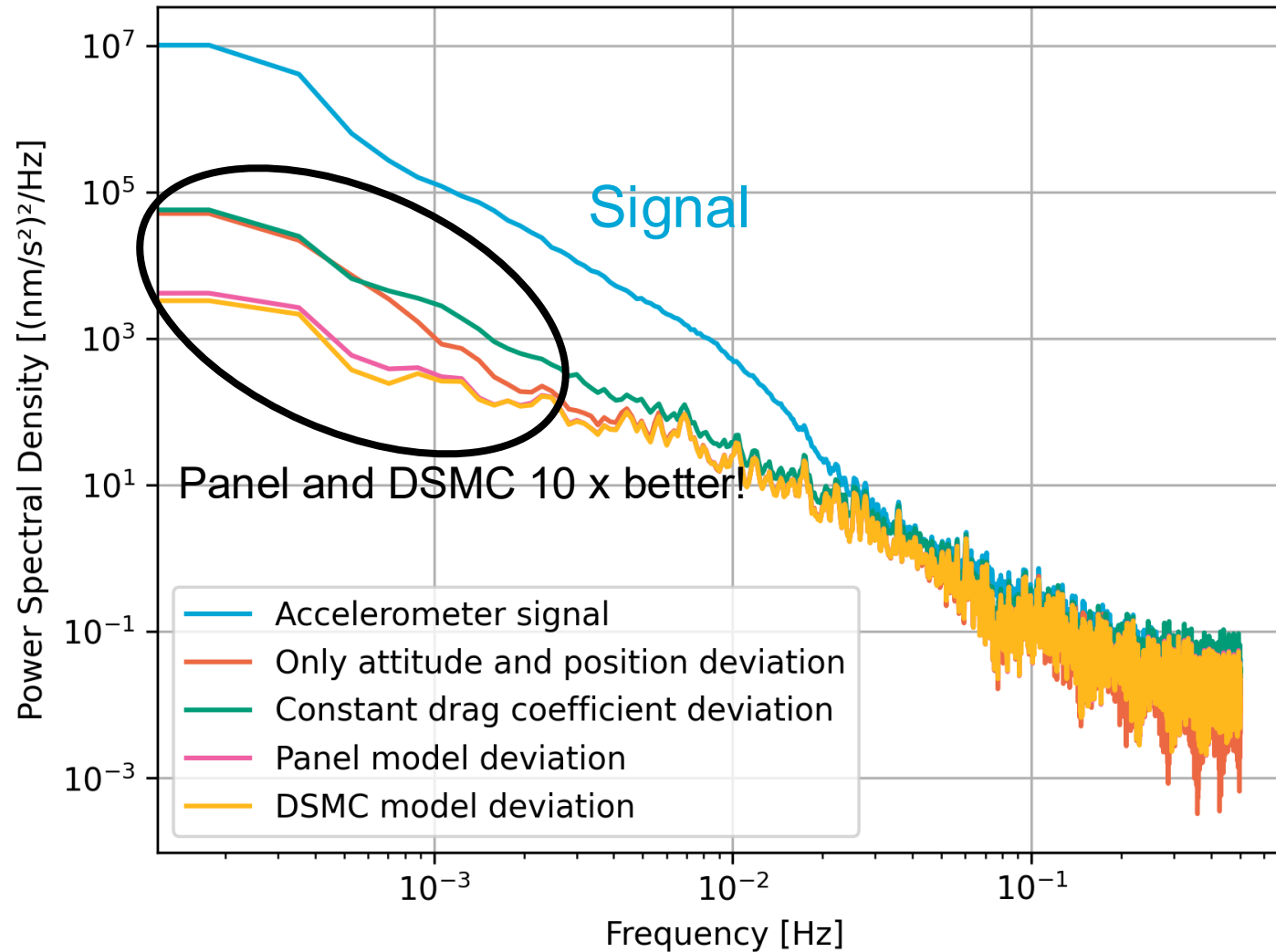
Case study: aerodynamic model LOS



The deviation between transplant and accelerometer



PSD – window 1 orbit



Conclusions and looking forward

Aerodynamic model matters!

- Constant drag coefficient performs worst
- Panel and DSMC perform (almost) equally well

Future research

- Radiation model effect
- Thruster firings
- Investigate the effect of improved transplant on gravity field models

Transplant software (Python) will be made open-source after first publication

References

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