

# Global Navigation Satellite System Occultation Sounder II (GNOS II)

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# Outline

◆1、GNSS RS Group, NSSC

◆2、Performance of FY-3C GNOS

◆3、Configuration of GNOS II

◆4、Improvements of GNSS RO module

◆5、GNSS-R module of GNOS II

◆6、Conclusions and Prospects

## ➤ Research domains:

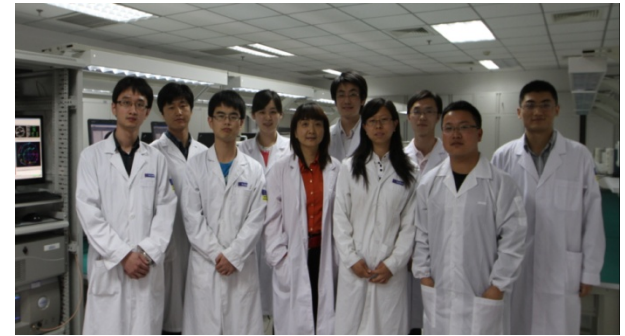
- GNSS radio occultation (GNSS RO)
- GNSS reflectometry (GNSS-R)
- LEO-LEO microwave occultation (LMO)

## ➤ Research Goals:

- To develop occultation and GNSS-R instruments
- To apply the GNSS remote sensing data

## ➤ Research partners:

- China Meteorological Administration (CMA)
- Institute of Atmosphere Physics (IAP), CAS
- Wegener Center for Climate and Global Change (WEGC)
- Geo Forschungs Zentrum (GFZ)
- Institut d'Estudis Espacials de Catalunya (IEEC)
- RMIT University
- ...



Researchers in GNSS RS Group

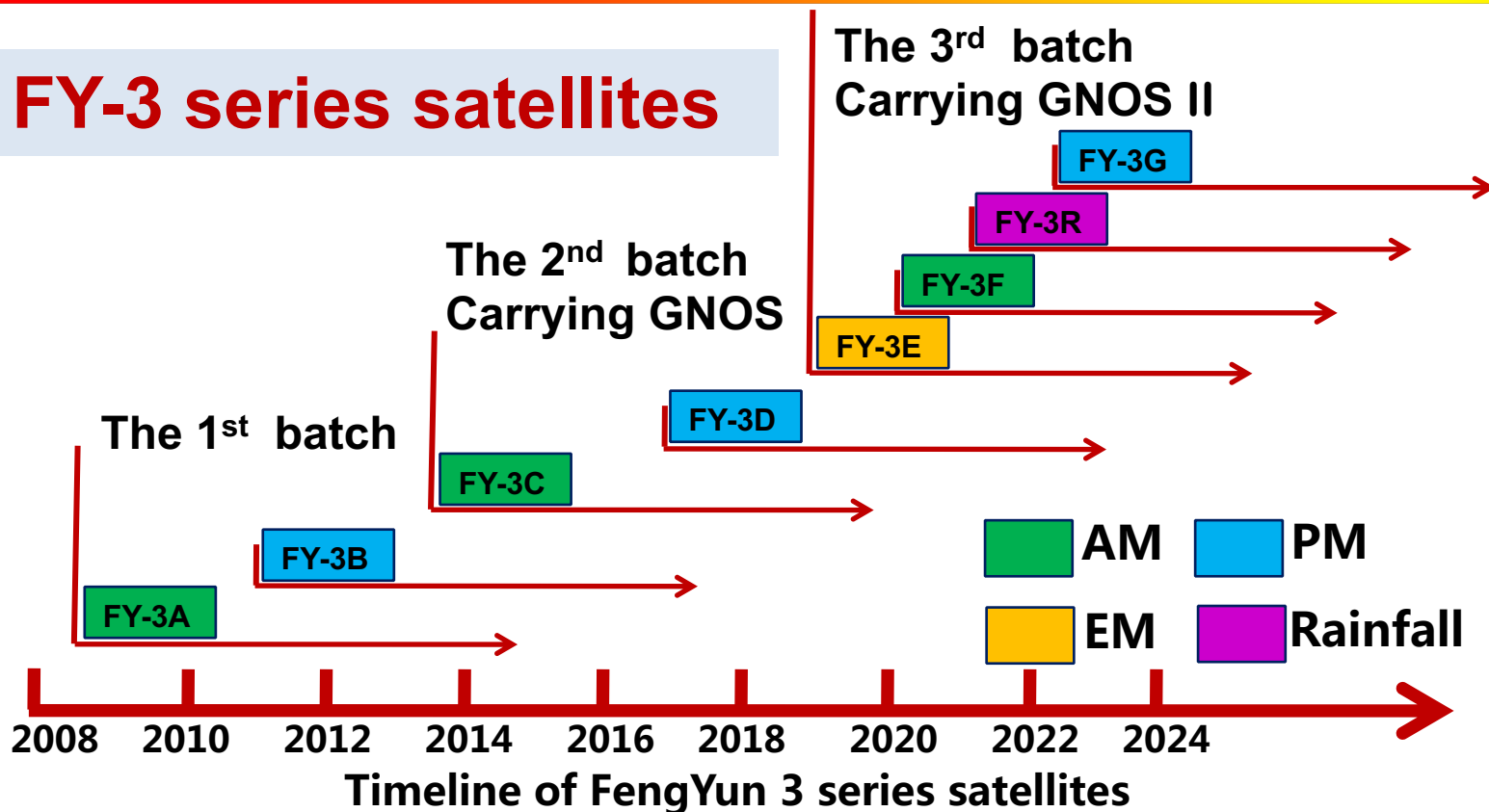


Researchers in GNSS RS Group and Prof Gottfried Kirchengast



Researchers in GNSS RS Group and Prof Martin-Neira

## FY-3 series satellites



- **FY-3C GNOS was launched on Sep 23 2013**
- **FY-3D GNOS will be launched in this Noveber**
- **FY-3E GNOS II plan to be launched in 2018**
- **The following FY-3 satellites will carry GNOS II as a key payload...**

## GNSS RO Receivers

GPS RO receivers

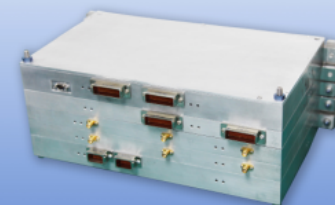


FY3 GPS/BDS compatible flight model

FY-3C/D GNOS

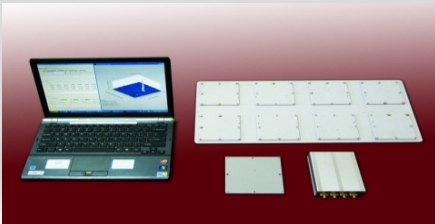


FY-3E GNOS II

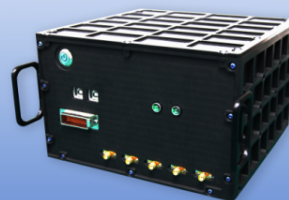


## GNSS Reflection Receivers

GPS-R software receiver



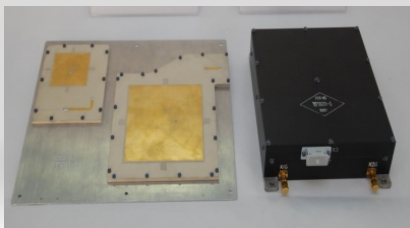
FPGA+DSP GPS/BDS/GALILEO compatible ocean reflection receiver



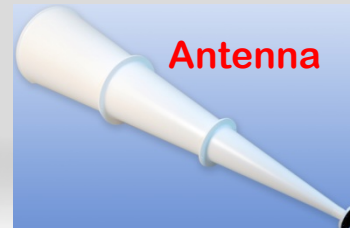
Main characters:  
 ●GPS/BDS/GAL for RO  
 ●GPS/BDS for GNSS-R  
 Status: Design

## LMO Transmitters and Receivers

YH-1 RO receiver flight model for Mars



LMO transmitter receiver for Earth



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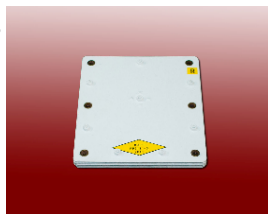
◆3、Configuration of GNOS II

◆4、Improvements of GNSS RO module

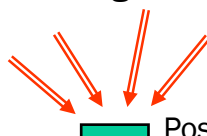
◆5、GNSS-R module of GNOS II

◆6、Conclusions and Prospects

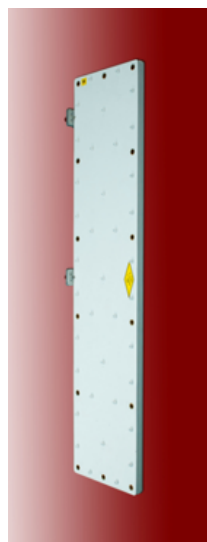
Positioning Antenna



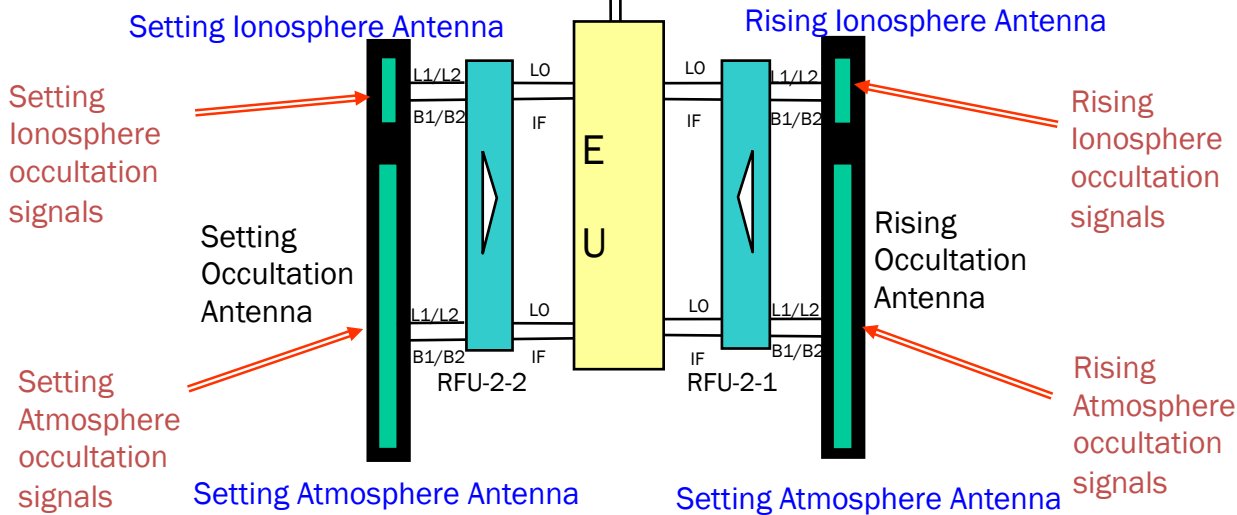
GNSS Navigation Signals



RFU-1



Setting Occultation Antenna



Setting Ionosphere occultation signals

Setting Occultation Antenna

Setting Atmosphere occultation signals

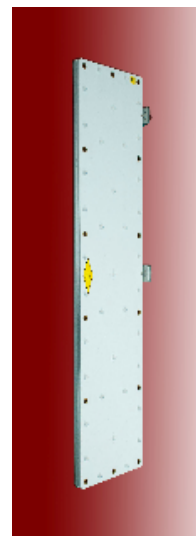
Setting Atmosphere Antenna

Setting Atmosphere Antenna

Rising Ionosphere occultation signals

Rising Occultation Antenna

Rising Atmosphere occultation signals



Rising Occultation Antenna

RFU-2-2



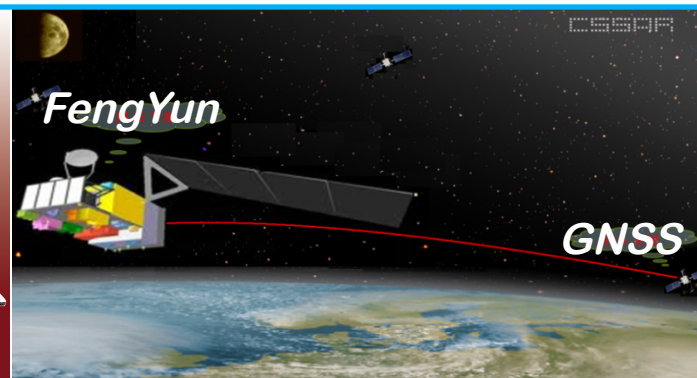
EU



RFU-2-1



## GNOS (GNSS Occultation Sounder)



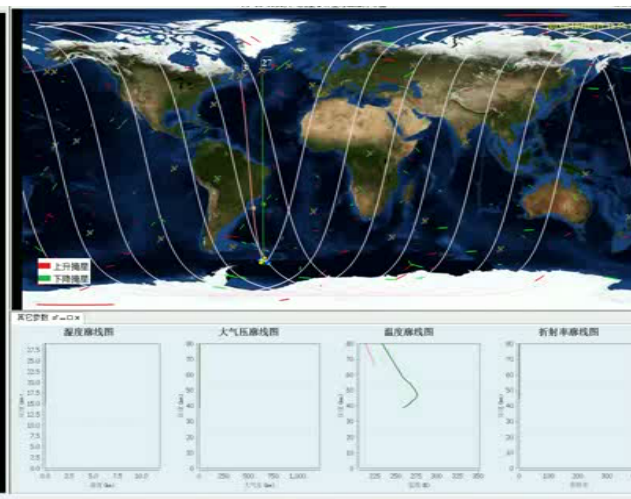
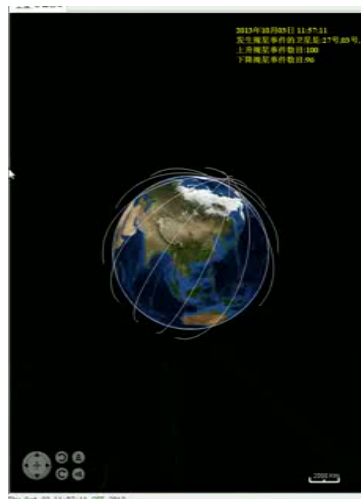
### FY-3C GNOS:

**Launched on 23 September 2013;  
Switched on at 21:48 (LT), on 29 Sep, 2013;**

**Received both GPS&BDS signals;  
>500 GPS RO events/day  
~200 BDS RO events/day**

**Main atm&ion RO retrievals:  
Refractivity (N), density( $\rho$ ),  
pressure (p), temperature (T),  
humidity (q), ionospheric  
electron density (Ne) et al.**

**(Thanks for funding from CMA)**



Statistics of the main atmosphere retrievals of FY-3C GNOS

0-30km	N(%) [mean/std]	T(K) [mean/std]	p(%) [mean/std]	q(g/kg) [mean/std]
GNOS vs Radio sound	-0.18/1.58	0.06/2.23	0.06/1.18	<0.3/<1.5
GNOS vs ECMWF	-0.14/0.98	-0.26/1.75	-0.18/1.05	<0.2/<1.2

Ref: M. Liao et al, Preliminary validation of the refractivity from the new radio occultation sounder GNOS/FY-3C, amt, 2016



Signal from the up-looking antenna are used for POD of the FY-3C satellite.

Tab.1 POD Precision Statistics of Position

	Radial RMS(cm)	Along RMS(cm)	Cross RMS(cm)	3D RMS(cm)
MEAN	0.919	1.460	2.232	2.868
MEDIAN	0.925	1.460	2.330	2.911

Tab.2 POD Precision Statistics of Velocity

	Radial RMS (mm/s)	Along RMS (mm/s)	Cross RMS (mm/s)	3D RMS (mm/s)
MEAN	0.018	0.014	0.009	0.025
MEDIAN	0.018	0.014	0.009	0.024

**GPS POD:**

3D RMS position error is **2.868 cm**

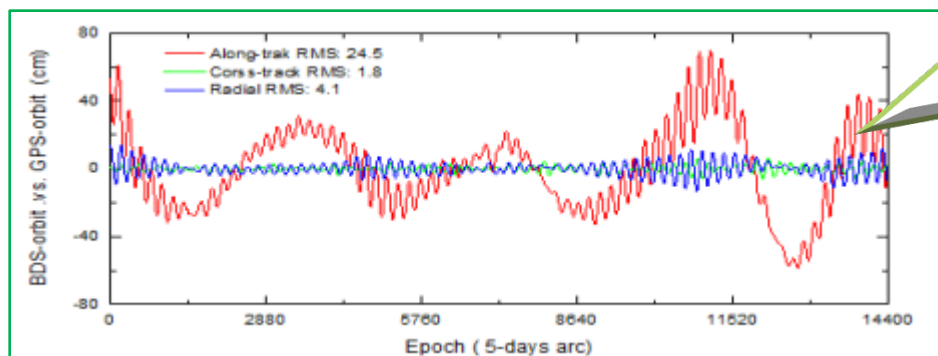
3D RMS velocity error is **0.025 mm/s**

**BDS POD :**

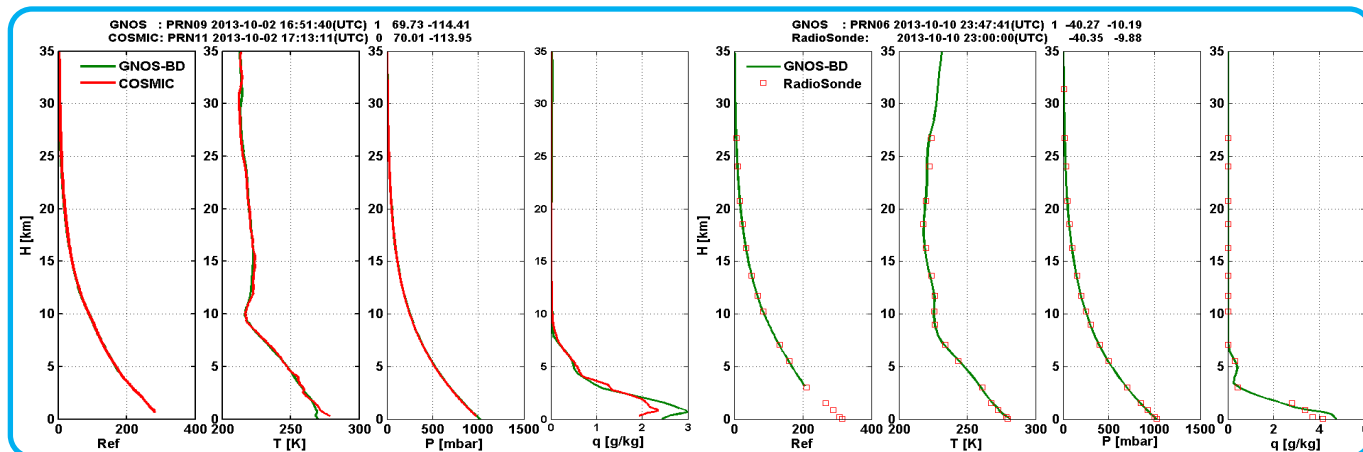
3D RMS position error is **30 cm**

**Error Sources:**

Regional coverage,  
BDS channel number,  
POD algorithm...

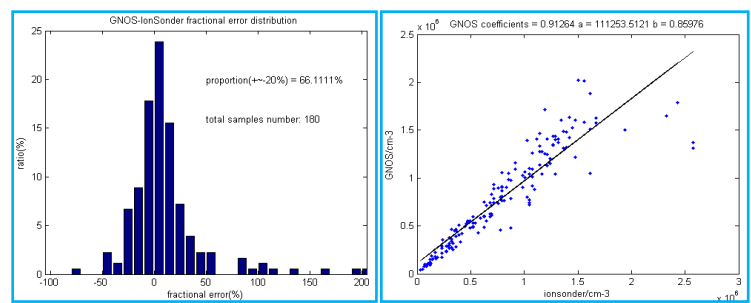
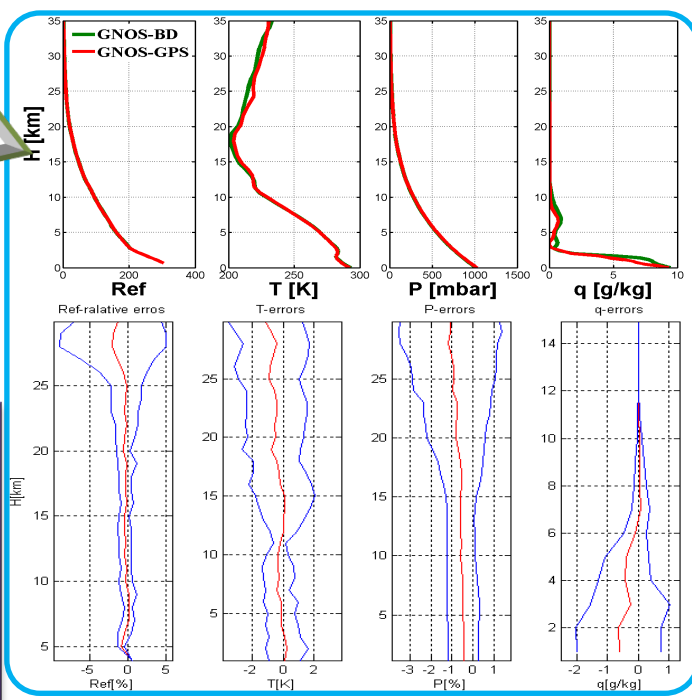


POD error using BDS ( 5 days)



Comparison with COSMIC observation data and radio sonde data

Comparison between GNOS/BDS and GNOS/GPS RO data



Ele. Density peak GNOS RO	Bias 2.8%	Std 18.3%	Correlation 0.91
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For the first time validated the consistence of GPS & BDS RO Data.

The quality of FY-3C GNOS RO data were consistent with other RO missions and other atmospheric observations.

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◆1、GNSS RS Group, NSSC

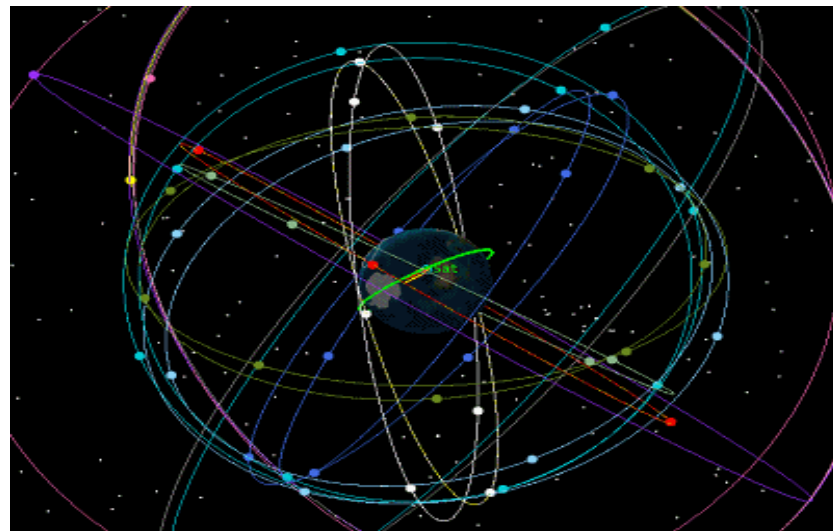
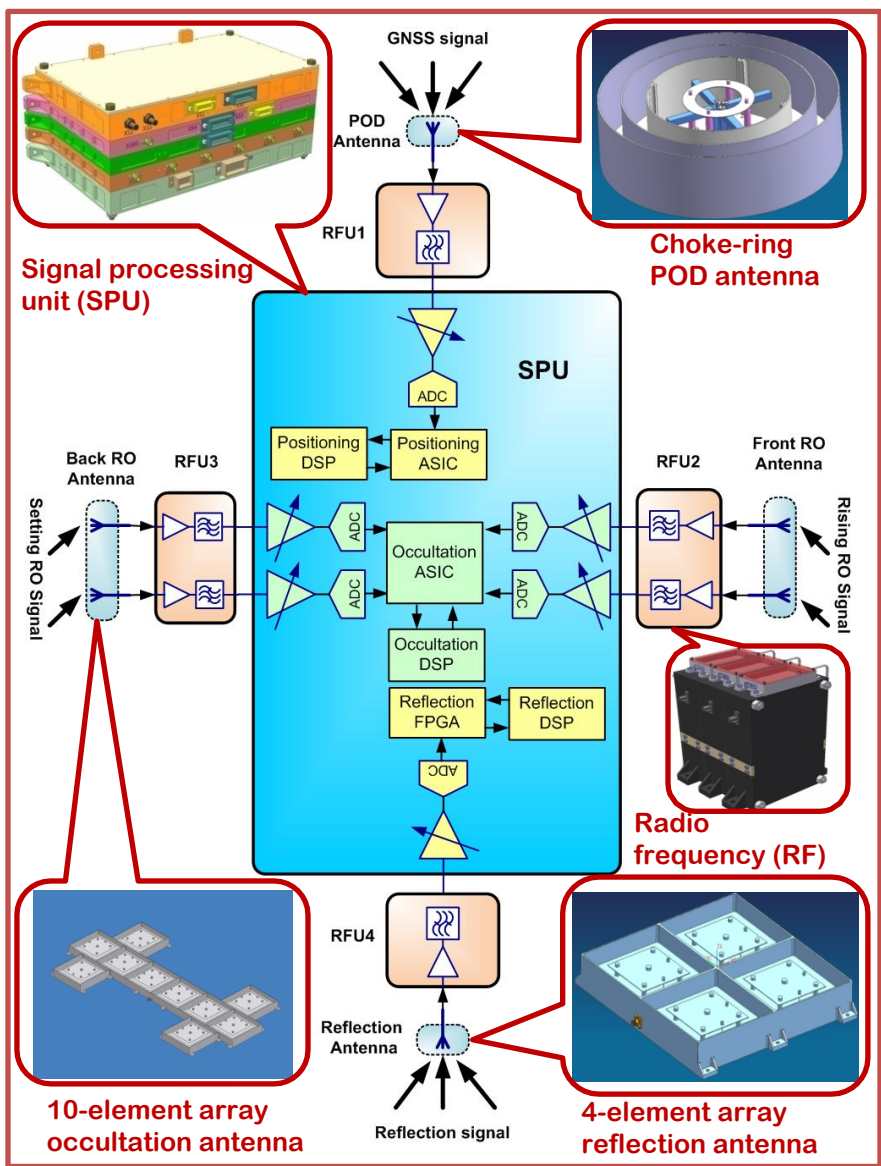
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Simulation of GNSS-R technique

**Main characters:**

- GPS/BDS/GAL for RO
  - GPS/BDS for GNSS-R
- Launch Satellite:**
- Altitude: 833km
  - Inclination: 98°
  - Launch time: 2018
- Status: Design**

To integrate the GNSS RO and GNSS-R in one payload.

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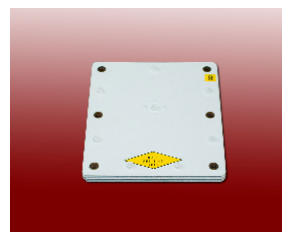
◆6、Conclusions and Prospects

## Improvements of GNSS RO module for GNOS II :

### ■ Enhancements of antenna:

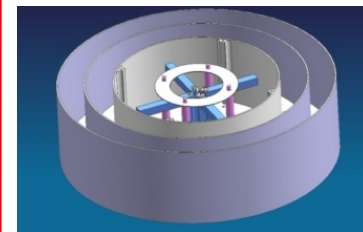
- A choke-ring POD antenna will be used to replace the patch POD antenna.
  - To reduce the backward and side radian of the antenna.
- 10-element array occultation antenna will be used to replace the 5-element occultation antenna.
  - To increase the occultation signals coverage and signal-to-noise ratio.

### FY-3C GNOS

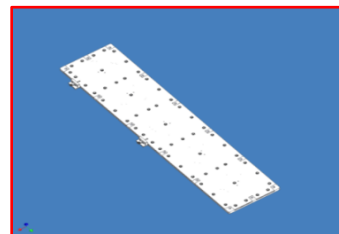
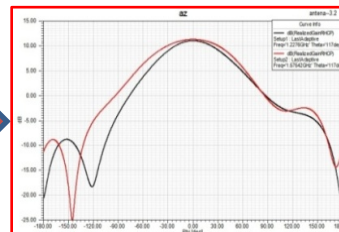


Plain POD antenna

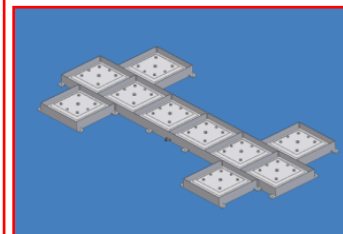
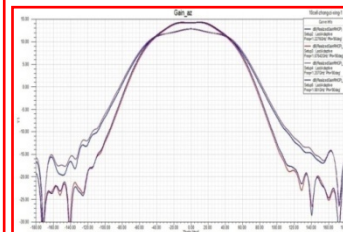
### FY-3E GNOS II



Choke-ring POD antenna



7.0 dBi@40° azimuth



10 dBi@45° azimuth

## ■ Increase of the channel numbers

- GPS, BDS and Galileo single channels for positioning and occultation.

- To achieve higher POD accuracy and more RO events

- Plus L2C and B1I open loop channels

- To improve the data quality of atmosphere near the Earth surface

## ■ Radiation tolerance ASIC

- 65-nm bulk silicon technology

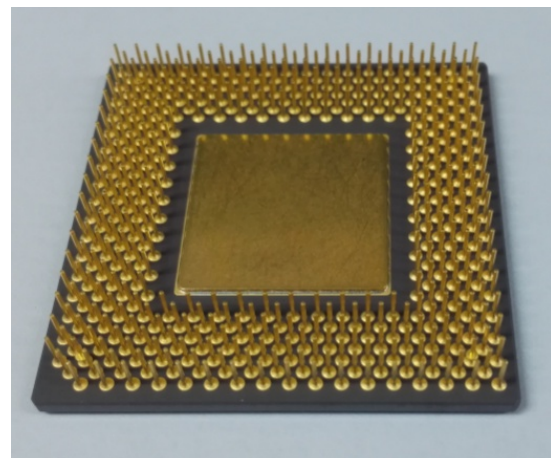
- SEU: 15 MeV·cm<sup>2</sup>/mg

- SEL: 75 MeV·cm<sup>2</sup>/mg

- TID: 100 krad (Si)

- To reduce the power consumption, volume and mass of the receiver.

GNOS	GNOS II
<b>14 GPS DF channels: (L1C/A+L2P)</b> 8 for position 6 for occultation	<b>26 GPS DF channels (L1C/A+L2C+L2P)</b> 12 for position 14 for occultation
<b>8 BDS DF channels: (B1I+B2I)</b> 4 for position 4 for occultation	<b>16 BDS DF channels: (B1I+B2I)</b> 8 for position 8 for occultation
	<b>6 GAL DF channels: (E1b+E5b)</b> 4 for position 2 for occultation



**Application Specific Integrated Circuit (ASIC)**

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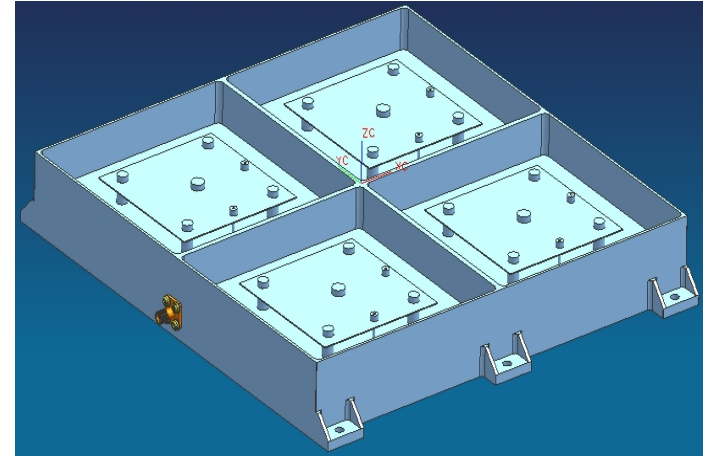
◆6、Conclusions and Prospects



## GNSS-R module parameters

The main primary parameters of GNSS-R module

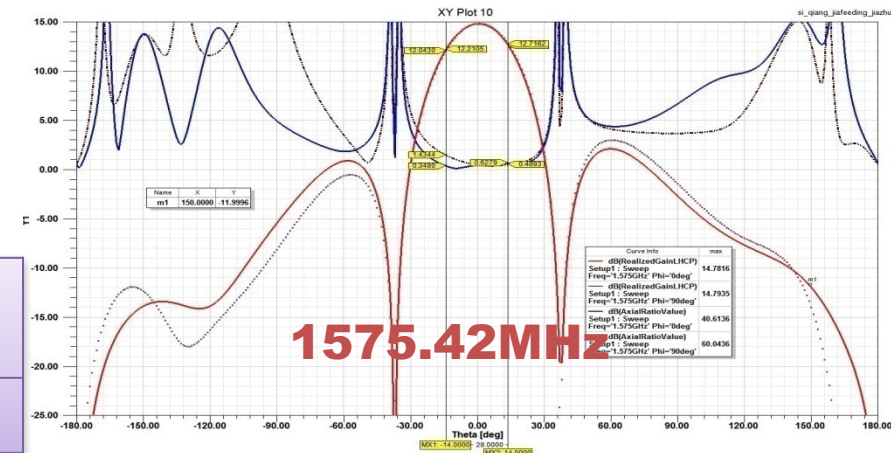
<b>Frequency</b>	<b>GPS: L1</b> <b>BDS: B1</b>
<b>Channel numbers</b>	<b>8</b>
<b>Sampling rate</b>	<b>61 MHz</b>
<b>Code tag number</b>	<b>122</b>
<b>Code tag interval</b>	<b>0.25 code chip</b>
<b>Code coverage</b>	<b>30.5 chip</b>
<b>Doppler tag number</b>	<b>24</b>
<b>Doppler tag interval</b>	<b>500 Hz</b>
<b>Doppler coverage</b>	<b>12 KHz</b>
<b>Output rate</b>	<b>750 Kbps</b>



The model of GNOS II GNSS-R antenna (LHCP)

The parameters of GNOS II GNSS-R LHCP antenna

<b>Frequency</b> (MHz)	<b>3dB BW</b> ( $\theta$ )	<b>3dB BW</b> ( $\varphi$ )	<b>Peak Gain</b> (dBi)
<b>1575.42</b>	<b>-16°~16°</b>	<b>-16°~16°</b>	<b>14.79</b>



Simulated antenna patterns by using ANSOFT HFSS

## Validation airborne campaign



**First flight:**  
 2015.11.11 9:30~11:30  
 Weather: cloudy/fog  
 wind :~1.5m/s  
 Height:500m and 2500m

Left circular polarization antenna, 4 dB

Thermal infrared imaging system

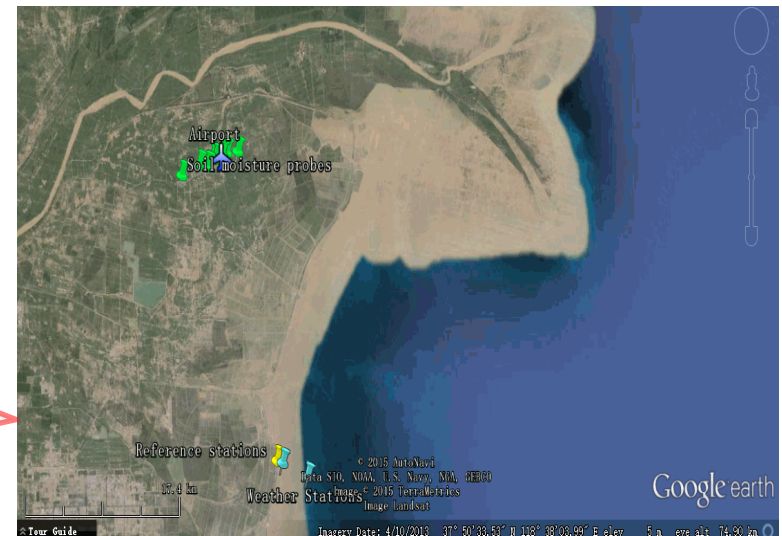
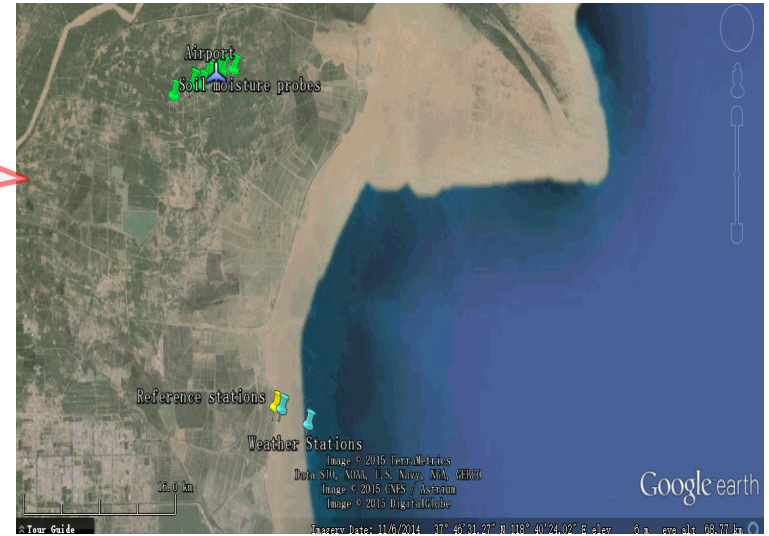


Right circular polarization antenna

Left circular polarization antenna, 13 dB

Air craft and GNSS-R antennas

**Second flight:**  
 2015.11.20 10:00~12:00  
 Weather: cloudy  
 wind: ~ 5.5m/s  
 Height: 500m



## Validation airborne campaign

The ground based in-situ observation stations and their measurements:

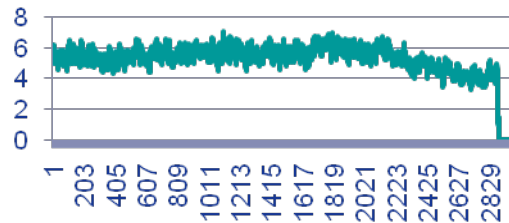


POD Reference Station

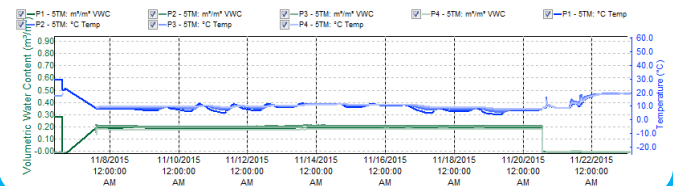


In-situ meteorology station

WS m/s



in-situ soil moisture station

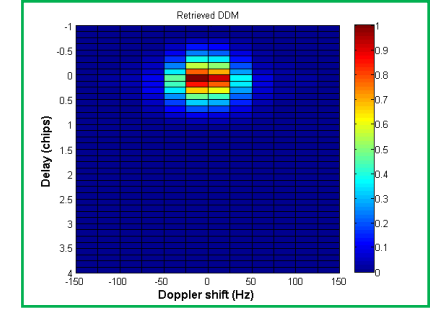
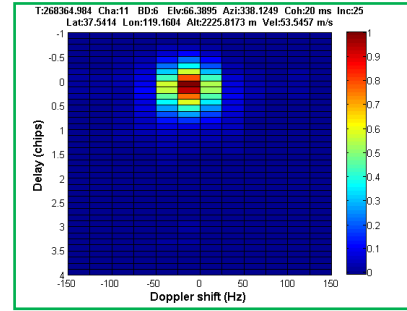
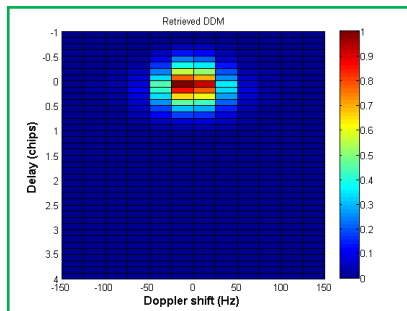
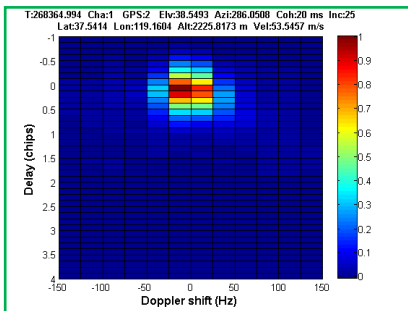
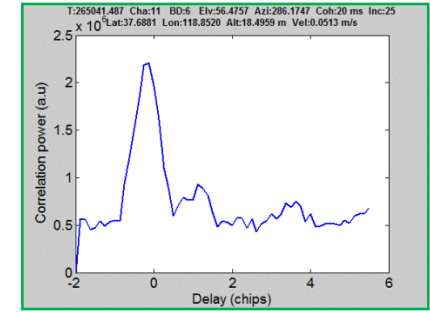
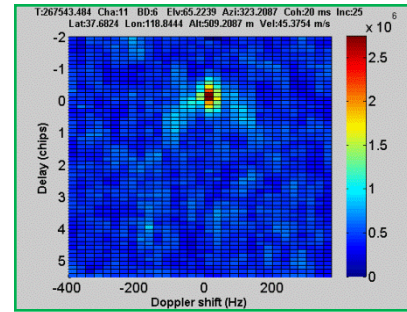
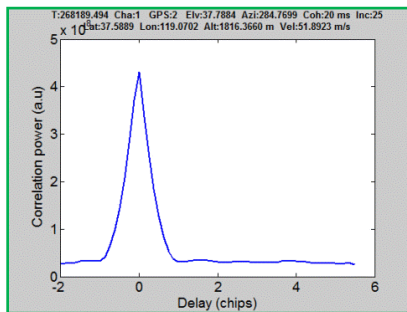
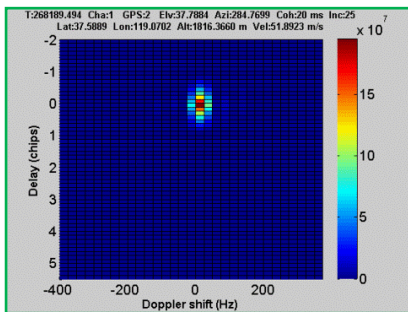


## Validation airborne campaign

### Preliminary results:

GPS:  $61 \times 32$  bins, 1/8 chip, 25 Hz

BDS:  $61 \times 32$  bins, 1/8 chip, 25 Hz



**Situ observed wind speed (average)**

**1.7 m/s**

**Retrieved wind speed (average)**

**2 m/s**

**Situ observed wind speed (average)**

**1.7 m/s**

**Retrieved wind speed (average)**

**2 m/s**

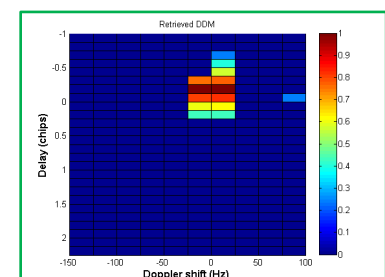
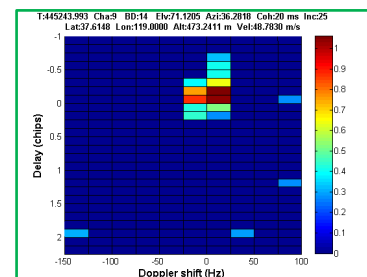
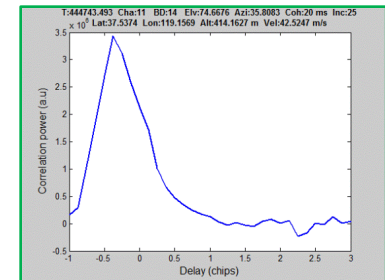
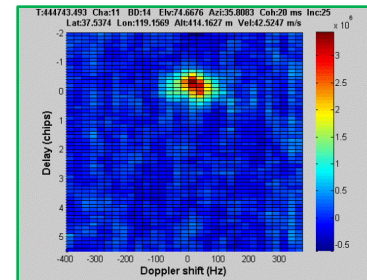
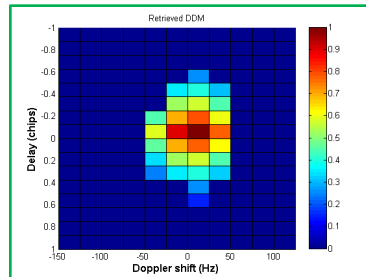
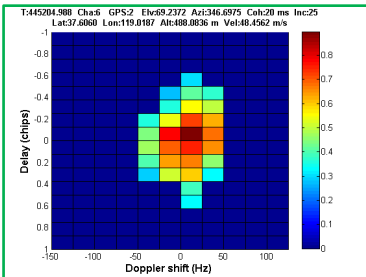
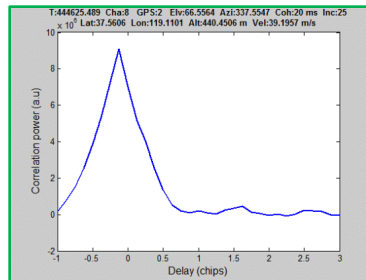
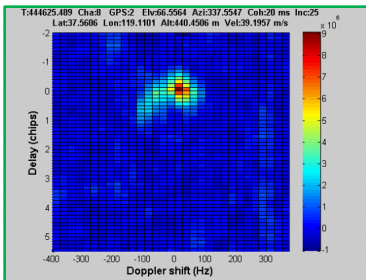
The campaign showed the consistence of the GNSS-R and in-situ observations.

## Validation airborne campaign

### Preliminary results:

GPS:  $61 \times 32$  bins, 1/8 chip, 25 Hz

BDS:  $61 \times 32$  bins, 1/8 chip, 25 Hz



**Situ observed wind speed (average)**

**5.6 m/s**

**Retrieved wind speed (average)**

**5 m/s**

**Situ observed wind speed (average)**

**5.6 m/s**

**Retrieved wind speed (average)**

**5 m/s**

The campaign showed the consistency of the GNSS-R and in-situ observations.

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- Several GNSS RO and GNSS R receivers have been designed and manufactured.
- FY-3D GNOS will be launched in this November.
- GNOS II is being designed and developed for FY-3E satellite, which will integrate the GNSS RO and GNSS R techniques in one payload. So far, these two modules have been mainly demonstrated by a space-based mission and an air-borne campaign, respectively.
- The following FY-3 satellites will carry GNOS II as a key payload.
  - ◆ Will the FY-3C GNOS work well when it's 5-year old?
  - ◆ Will the FY-3D GNOS perform better than the FY-3C GNOS?
  - ◆ Will the FY-3E GNOS II go ahead well in 2018?
  - ◆ What is the status and new results of other missions worldwide? (COMIS[-2], MetOp, CYGNSS, research missions, commercial missions... )
  - ◆ Welcome to Beijing to find them out via a conference.

# LSEE NSSE Preliminary Conf. information

- Meeting plan: **IGL-1 2018**—Welcome!
- The first **Innovations in GNSS and LEO Occultations & Reflections Workshop**
- Date: September 2018 Location: Beijing (**TBC, offer joint with IROWG-7, to be coordinated with IROWG, ICGPSRO meetings**)



Landscapes  
in Beijing





Thank you!